

## Semianalytical Computation of Heteroclinic Connections Between Center Manifolds with the Parameterization Method\*

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**Abstract.** This paper presents a methodology for the computation of whole sets of heteroclinic connections between isoenergetic slices of center manifolds of center  $\times$  center  $\times$  saddle fixed points of autonomous Hamiltonian systems. It involves (a) computing Taylor expansions of the center-unstable and center-stable manifolds of the departing and arriving fixed points through the parameterization method, using a new style that uncouples the center part from the hyperbolic one, thus making the fibered structure of the manifolds explicit; (b) uniformly meshing isoenergetic slices of the center manifolds, using a novel strategy that avoids numerical integration of the reduced differential equations and makes an explicit three-dimensional representation of these slices as deformed solid ellipsoids; (c) matching the center-stable and center-unstable manifolds of the departing and arriving points in a Poincaré section. The methodology is applied to obtain the whole set of isoenergetic heteroclinic connections from the center manifold of  $L_2$  to the center manifold of  $L_1$  in the Earth-Moon circular, spatial restricted three-body problem, for nine increasing energy levels that reach the appearance of halo orbits in both  $L_1$  and  $L_2$ . Some comments are made on possible applications to space mission design.

**Key words.** parameterization method, heteroclinic connections, invariant tori, libration point orbits, RTBP, center manifolds

**MSC codes.** 37C29, 37N05, 37M05, 70F07

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**1. Introduction.** Heteroclinic connections play an important role in the description of dynamical systems from a global point of view. In this work we address the systematic computation of heteroclinic connections between center manifolds of fixed points of center  $\times$  center  $\times$  saddle type of autonomous, 3-degrees-of-freedom Hamiltonian systems.

Our interest in this case comes from applications in astrodynamics, namely to libration point space missions. In these missions, spacecraft are sent to orbits that stay close to the fixed (in a rotating frame) points  $L_1$ ,  $L_2$  of the spatial, circular restricted three-body problem (RTBP). This model describes the dynamics of a massless particle under the attraction of

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